Side branch resonators modelling with Green's function methods

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Introduction

Side branch resonators are commonly used for engine exhaust noise control : (i) low-frequency applications with a single plane wave mode (automotive) (ii) medium-frequency applications and highly multimodal context (aeronautics).



Figure: Side branch resonator (cavity Ω) with two openings Γ .

Numerical predictions : (i) 1D approximations (with length corrections); (ii) FEM, BEM -> computationally demanding (this includes mesh preparation etc...) especially in a highly multimodal context, lack of physical interpretation.

In the frequency domain, the acoustic pressure must obey the Helmholtz equation

$$\Delta p + k^2 p = 0, \quad k = \omega/c,$$

and $q = \partial_n p = 0$ everywhere *except* on Γ . The Green's function for the rigid-wall cavity is given by the infinite series

$$G_{\Omega}(\mathbf{r},\mathbf{r}') = \sum_{n=0}^{\infty} \frac{\phi_n(\mathbf{r})\phi_n(\mathbf{r}')}{\lambda_n - \lambda}$$

where $\lambda = \omega^2$. Eigenfunctions ϕ_n are properly normalized so that application of the Green's theorem in the cavity yields

$$p(\mathbf{r}) = \int_{\Gamma} G_{\Omega}(\mathbf{r}, \mathbf{r}') \, q(\mathbf{r}') \, \mathrm{d}\Gamma(\mathbf{r}')$$

Impedance matrix

We need to precompute a finite set of eigenfunctions and estimate the truncation error...

Consider the eigenvector Φ_n , the FE discretization of *n*th eigenmode ϕ_n :

$${f A}(\lambda_n) \Phi_n = 0$$
 where ${f A}(\lambda) = {f K} - \lambda {f M}$

After reduction to the interfacial nodes, we obtain the impedance matrix

$$\mathsf{Z}(\lambda) = \mathsf{I}_{\mathsf{\Gamma}}^{\mathrm{T}} \, \mathsf{A}^{-1}(\lambda) \, \mathsf{I}_{\mathsf{\Gamma}}$$

with

$$\mathbf{A}^{-1}(\lambda) = \sum_{n=0}^{\infty} \frac{\Phi_n \Phi_n^{\mathrm{T}}}{\lambda_n - \lambda} = \Phi \, \mathbf{D}(\lambda) \, \Phi^{\mathrm{T}}$$

Finally,

$$\tilde{\mathbf{p}} = \mathbf{Z}(\lambda) \, \tilde{\mathbf{F}} \, \mathbf{q} = \mathbf{G}_{\Omega} \, \mathbf{q}$$

Truncation

Keeping the first N eigenmodes gives

$$\mathbf{Z}(\lambda) = (\tilde{\Phi} \, \mathbf{D}(\lambda) \, \tilde{\Phi}^{\mathrm{T}})|_{N} + \mathbf{R}(\lambda).$$

The correction term ${\bf R}$ remains weakly dependent on the frequency, so we can take the first order Taylor expansion

$$\mathbf{R}(\lambda) \approx \mathbf{R}(\overline{\lambda}) + (\lambda - \overline{\lambda}) \frac{\partial \mathbf{R}}{\partial \lambda} + \dots$$

The residual matrices are computed via

$$\mathbf{R} = \mathbf{I}_{\Gamma}^{\mathrm{T}} \mathbf{A}^{-1} \mathbf{I}_{\Gamma} - (\tilde{\Phi} \mathbf{D}(\lambda) \tilde{\Phi}^{\mathrm{T}})|_{N} \partial_{\lambda} \mathbf{R} = \mathbf{I}_{\Gamma}^{\mathrm{T}} \mathbf{A}^{-1} \mathbf{M} \mathbf{A}^{-1} \mathbf{I}_{\Gamma} - (\tilde{\Phi} \partial_{\lambda} \mathbf{D} \tilde{\Phi}^{\mathrm{T}})|_{N}$$



Scattering matrix

The theory starts by introducing the hard-walled duct Green's function

$$G(\mathbf{r},\mathbf{r}') = \sum_{l=0}^{\infty} \frac{\psi_l(x,y)\psi_l^*(x',y')}{2\mathrm{i}\beta_l} \,\mathrm{e}^{\mathrm{i}\beta_l|z-z'|}$$

The transverse eigenmodes ψ_l are solution of the boundary value problem

$$(\partial_{xx}^2 + \partial_{yy}^2)\psi_I + k^2\psi_I = \beta_I^2\psi_I$$

with $\partial_n \psi_l = 0$ on the boundary line ∂S . These modes are normalized as

$$\int_{\mathcal{S}} |\psi_I|^2 \,\mathrm{d}\mathcal{S} = 1$$

in particular $\psi_0 = 1/\sqrt{A_d}$ where A_d is the cross section area of the main duct. For circular ducts :

$$\psi_l = N_{m,n} J_m(\alpha_{m,n} r) \mathrm{e}^{\mathrm{i} m \theta}, \quad \beta_l = \sqrt{k^2 - \alpha_{m,n}^2}$$

Scattering matrix

The finite element discretization of the integral equation

$$p(\mathbf{r}) = \int_{\Gamma} G(\mathbf{r}, \mathbf{r}') q(\mathbf{r}') d\Gamma(\mathbf{r}') + p'(\mathbf{r})$$

gives (\mathbf{r}_i is the FE node location)

$$ilde{p}_i = \sum_{j=1}^{ ilde{N}} G_{ij} q_j + p_i^{\prime} \quad ext{with} \quad G_{ij} = \int_{\Gamma} G(\mathbf{r}_i, \mathbf{r}^{\prime}) \, ilde{\phi}_j(\mathbf{r}^{\prime}) \, \mathrm{d}\Gamma(\mathbf{r}^{\prime})$$

The acoustic velocity is deduced from

$$\mathbf{q} = (\mathbf{G}_{\Omega} - \mathbf{G})^{-1} \, \mathbf{p}'$$

Note :

i. The computation is not trivial as $\partial_{z'} G$ is discontinuous at $z' = z_i$.

ii. The matrix $\mathbf{G}_{\Omega} - \mathbf{G}$ is of small size.

Simplified models : one opening

Starting with one opening only, the impedance matrix relation can be averaged to give

$$ar{p} = ar{Z}(\lambda)ar{q}$$
 where $ar{Z}(\lambda) = rac{1}{ ilde{N}} \sum_{i=1}^{ ilde{N}} \sum_{j=1}^{ ilde{N}} (\mathbf{Z}(\lambda)\, ilde{\mathbf{F}})_{ij}$

By the same token,

$$ar{p} = rac{1}{W}ar{q} + ar{p}', \hspace{1em} ext{where} \hspace{1em} W = rac{2 ext{i}kA_d}{A}$$

and A denotes the area of the interface. An incident plane wave $p^{l} = e^{ikz}$ produces a transmitted pressure field

$$p = T \mathrm{e}^{\mathrm{i}kz}$$
 with $T = 1 + rac{1}{W ar{Z}(\lambda) - 1}$

Helmholtz resonators



Figure: Classical (left); with extended neck (right).

Helmholtz resonator, classical



Helmholtz resonator, with extended neck



Simplified models : two openings

We consider a symmetric resonator connected to the main duct via two openings located at $z = z_1$ and $z = z_2$.

$$\left(\begin{array}{c} \bar{p}_1\\ \bar{p}_2 \end{array}\right) = \left(\begin{array}{c} \bar{Z}_{11} & \bar{Z}_{12}\\ \bar{Z}_{12} & \bar{Z}_{11} \end{array}\right) \left(\begin{array}{c} \bar{q}_1\\ \bar{q}_2 \end{array}\right)$$

Moreover,

$$\left(\begin{array}{c} \bar{p}_1\\ \bar{p}_2 \end{array}\right) = \frac{1}{W} \left(\begin{array}{c} 1 & \mathrm{e}^{\mathrm{i}kL}\\ \mathrm{e}^{\mathrm{i}kL} & 1 \end{array}\right) \left(\begin{array}{c} \bar{q}_1\\ \bar{q}_2 \end{array}\right) + \left(\begin{array}{c} \bar{p}_1'\\ \bar{p}_2' \end{array}\right)$$

where $L = |z_2 - z_1|$. This gives

$$T = 1 + \frac{2W\bar{Z}_{11} - 2W\bar{Z}_{12}\cos(kL) + (e^{2ikL} - 1)}{(W\bar{Z}_{11} - 1)^2 - (W\bar{Z}_{12} - e^{ikL})^2}$$

Thus, no acoustic energy is transmitted if

$$\bar{Z}_{12}^2 - \bar{Z}_{11}^2 - \frac{A\sin(kL)}{kA_d}\bar{Z}_{12} = 0$$

Herschel-Quincke resonator



Herschel-Quincke resonator





The fan noise is one of the dominant components at take-off and landing for aircraft with modern high bypass ratio turbofan engines : broadband noise + Blade Passing Frequency (BPF) tones

Fan noise

Actual configuration...



Model...



Validation on a small size configuration



	Matrix size	CPU time (Matlab)
Our model	500	1 h 50 min
FEM	82 000	31 h 15 min

Optimal configuration (36 HQ tubes)



- ► Incident
- ► Liner-HQ system

What does the liner do?



Influence of the number of tubes on the first BPF (iso-surface)



The proposed Green's function based method allows to reduce the computational effort as only the acoustic velocity at the interface needs to be calculated.

A very high number of propagative modes (few hundreds) can be handled easily on a single PC.

Gives access to physical interpretation in the low-frequency regime.

In prospect : - could be used for designing taylor-made resonators using optimization procedures. - viscosity effects should be included